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|-------|-------|-------|-------|------|---|
|       | FOR   | FAA   | USE   | ONLY | _ |
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## MAJOR REPAIR AND ALTERATION Copo (Airframe, Powerplant, Propeller, or Appliance)

| INSTRUCT                   | IONS: Print or typons and disposition                           | e all entries. See I<br>of this form.   | FAR 4                | 3.9,                        | FAR 43 Appe                                  | ndix B, and                                    | AC 43.9-1 (or :         | subsequent i                | evision th                                       | ieteol)            |
|----------------------------|---|---|----------------------|-----------------------------|--|--|-------------------------|-----------------------------|--|--------------------|
|                            | MAKE  | m o w   |                      |                             |  | MODEL  | A 22-150                |                             |  | 1                  |
| 1. AIRCRAFT                | Piper   |   |                      |                             |  | NATIONALITY AND REGISTRATION MARK              |                         |                             |  |                    |
|                            | 22-5 <b>2</b> 50  |   |                      |                             |  | ท7522D   |                         |                             |  |                    |
|                            | NAME (As shown on registration certificate)                     |   |                      |                             |  | ADDRESS (As shown on registration certificate) |                         |                             |  |                    |
| 2. OWNER                   | Thomas  | D. Stewart  |                      |                             | į  | 405 Norman Ave.<br>Cashmere, Wn. 98815         |                         |                             |  |                    |
| 3. FOR FAA USF ONLY        |   |   |                      |                             |  |  |                         |                             |  |                    |
|                            | (*)   | ia Trempor<br>Licenson in Lagra<br>Licenson in Lagra<br>Licenson in Lagra<br>Licenson in Lagra<br>B-11-86 | picker).<br>Naci i s | ر<br>آرائیون                | u iú Cultar                                  | ila irakening<br>Sa mira iai                   | 6 111E<br>4 64 <b>A</b> |                             |  |                    |
|                            |   | Date  | IT INC               | NTIC                        | FAA MSDECKEL                                 |  | <del></del>             |                             |  | TYPE               |
|                            | I   | 4. UNIT IDENTIFICATION  |                      |                             |  | SERIAL HO.                                     |                         | HO.                         | <del>-                                    </del> | ALTER-             |
| AM TIMU                    |   | KE  |                      |                             | WODEL  |  | SERIAL NO.              |                             | REPAIR   | ATION              |
| AIRFRAME                   | (As described in item 1 above)                                  |   |                      |                             |  |  |                         |                             | xx   |                    |
| POWERPLANT                 |   |   |                      |                             |  |  |                         |                             |  |                    |
|                            | <del> </del>  |   |                      |                             |  |  |                         |                             | <u> </u>   | <del>  </del>      |
| PROPELLER                  |   |   |                      |                             |  |  | *=                      |                             |  |                    |
|                            | TYPE  |   |                      |                             |  |  |                         |                             | ļ  |                    |
| APPLIANCE                  | MANUFACTURER  |   |                      |                             |  |  |                         |                             |  |                    |
|                            |   |   |                      |                             |  |  |                         |                             | <u></u>  | <u> </u>           |
|                            |   |   | 6. (                 | CONF                        | DRAITY STATES                                |  | 100                     | C C55                       | TIFICATE   | NO.                |
| A.                         | AGENCY'S NAME   | AND ADDRESS   |                      | x                           |  | ID OF AGE                                      |                         |                             |  |                    |
|                            | William C.  | Markey  |                      |                             |  | FICATED MECHANIC ASP                           |                         |                             | 74008  | 6                  |
| 1506 Walla Walla St.       |   |   |                      | CERTIFICATED REPAIR STATION |  |  |                         |                             |  |                    |
|                            | Wenatchee,  | Wn. 98801   |                      |                             | MANUFACTURE                                  | l  |                         |                             |  |                    |
|                            | that the repair an<br>ents hereto have be<br>the information fu | en made in accorda  | ance w               | vith.                       | the requireme                                | nts of Fact 4                                  | 13 OL MIE (J.J. 1       | described o<br>Jederal Avia | n the tev<br>tion Regi                           | erse or<br>Harions |
| DATE                       | 4   |   |                      |                             | NATURE OF                                    |  |                         | <u>.</u>                    |  |                    |
| 3-11-86                    |   |   |                      | William Marky               |  |  |                         |                             |  |                    |
|                            |   |   |                      |                             | FOR RETURN T                                 |  |                         |                             |  | had by             |
| Pursuant to<br>the Adminis | the authority giver   | s persons specified<br>Il Aviation Adminit  | pelow                | n an                        | d is APP                                     | ROVED  | REJECTED                | 1) 1116 11141111            |  |                    |
| FA                         | RT. STANDARDS   | MANUFACTURER  | X                    | INS                         | PECTION AUTHOR                               | ZATION   | OTHER (Specify)         |                             |  |                    |
|                            | A DESIGNEE  | REPAIR STATION  |                      | Of<br>Of                    | NADIAN DEPARTM<br>TRANSPORT INSP<br>AIRCRAFT | ECTOR  |                         |                             |  |                    |
| REJECTION                  |   |   |                      |                             |  |  |                         |                             |  |                    |
| 3-//                       | 1-84  | 1)4008  | 10                   |                             | coul   | com  | <u> </u>                | W75                         |  | (8320)             |

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

March 7, 1986

Fabricated fuel cap covers from early Cessna 180 gas cap covers. These form a resevoir to catch and drain fuel spills overboard during refueling and prevent fuel siphoning from fuel caps. The original fuel neck and gas cap are unaltered. These units cover the existing cap to streamline them and to keep cap covered from the elements.

 The gas cap covers are early Cessna 180 covers and have the hump in the cover lid. The later flat style covers will not work as they don't allow clearance for the Piper gas caps. The Cessna covers are designed to attach to the bladder type fuel cell and will have to be modified as follows:

- 1. Remove Piper gas cap and measure distance from top of metal tank bay cover to top of gas tank. (probably about 5/8")
- 2. Remove tank bay cover from wing.
- 3. Take a piece of .032 or such flat sheet of aluminum and cut to about 10" sq. (should be about 2" larger than the dia. of Cessna cover.)
- 4. Mark center of alum. sheet described in #3 and cut a hole slightely larger than gas tank filler neck in flat sheet.
- 5. Now slide this piece down over filler neck and let it rest on top of gas tank. Put gas cap back on fuel tank.
- 6. Take Cessna cover and measure down from bottom side of top mounting ring on Cessna cover the distance measured in step #1. Mark this distance around neck of cover and cut the lower portion of cover off. This will leave about 5/8" sides on cover. The lower mounting ring which used to bolt to bladder tank is discarded.
- 7. With lid to Cessna cover latched closed, place cover over Piper gas tank filler neck. The Piper gas cap should center into the hump on Cessna cover. Push down until sides of cover contact the flat alum. plate laying on top of Piper gas tank. Should take very little pressure to do this unless sides of Cessna cover were cut to long. If so trim as necessary. The hump in Cessna cover is not exactly in center of cover so it is necessary to do this step to line things up.
- 8. Now draw a line around base of Cessna cover onto alum. sheet and make 2 or 3 index marks so when pieces are removed they can be accurately lined up for welding.
- 9. Remove parts from Piper tank. Cut alun. sheet as marked in previous step. Sides of Cessna tank cover should probably be trimed about 1/16" so finished unit does not jam down tight against gas tank when installed. Carefully align parts and weld prepared alum sheet to botton of Cessna cover. This forms the new botton of Cessna cover and if done properly the filler neck hole will line up exactly with the hump in Cessna cover. Also install a 1/8" piece of alum, tube on latch side of cover for fuel overflow drainage. This should be parallel to bottom and stick out about 2".

- 10. Install Piper tank bay cover on wing.
- 11. Slide modified Cessna cover down over filler neck and rotate cover so hinge line on cover is toward leading edge of wing. Mark Piper tank bay cover using outside bottom of Cessna cover for a pattern.
- 12. Remove Piper tank bay cover and cut out area previously marked.
- 13. Install Piper cover on wing and slip Cessna cover in place. Mounting ring on Cessna cover should contact top of Piper tank bay cover. Mark Piper tank bay cover for location of Cessna cover mounting holes.
- 14. Remove Cessna cover and Piper tank bay cover. Drill mounting holes in Piper cover and install correct size tinnerman nuts. Cessna cover can now be nounted to Piper tank bay cover.
- 15. Take approx. 18" of 1/8" alum, tubing and pass one end of it between rear spar and rear edge of gas tank straight down until it contacts lower fabric surface of wing. Mark this spot and cut out for 1/8" tubing to exit bottom of wing. Let it stick down 1/4" or so below bottom of wing. Bend remaining tubing forward to follow contour of fuel tank. The alum, tubing should be attached to rear spar with a small adel clamp etc. to secure it in place. Some duct tape works OK for on top of tank.
- 16. The tubing in #15 should be adjusted in length so there is about 8" or so between the end of it and the end of the tube from Cessna cover when cover is in place. A piece of rubber gas line hose connects these together and is a necessary coupling for installing cover. There is no need to clamp this hose.
- 17. Glue a piece of 12" sq. felt to top of gas tank to prevent rubbing (1/16"felt) of parts and install Piper tank bay cover with Cessna cover mounted in place.
- 18. Replace gas cap and close lid on Cessna cover. Coveriwill probably touch top of gas cap, but this is not a problem.

NOTE: I seal the area between filler neck and bottom of cover with a foam ring of rubber glued only to bottom of cover. If you do this it is very important to drill a couple of 1/16" vent holes in the top of Piper gas cap. If this is not done and rubber moves up filler neck it will block vents on bottom of Piper caps. Don't neglect to do this as I speak from experience. Also install good tight gas cap gaskets to prevent fuel slopping out around cap when tanks are full. THIS COMPLETES MODIFICATION